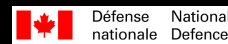
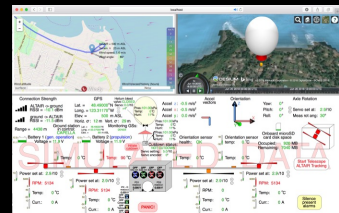
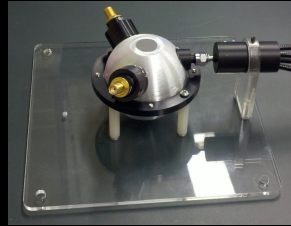
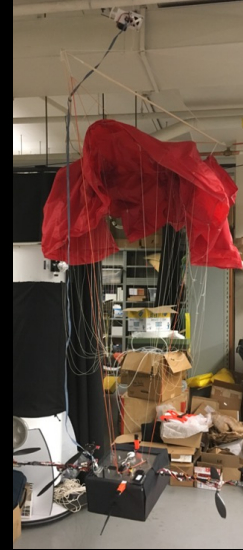
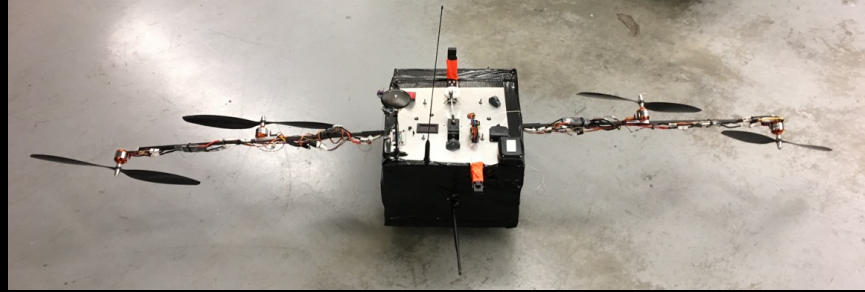
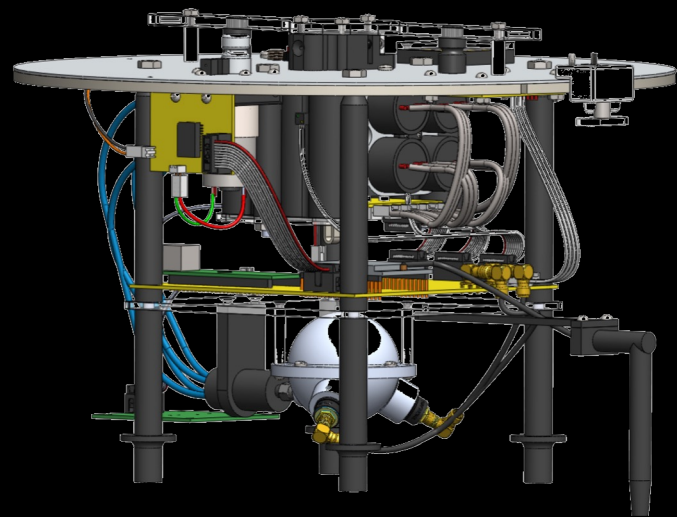


Dark Energy ... and a New Miniature Stratospheric Platform for Ultra-Low-Cost Communication, Precision Calibration, & Earth Observation



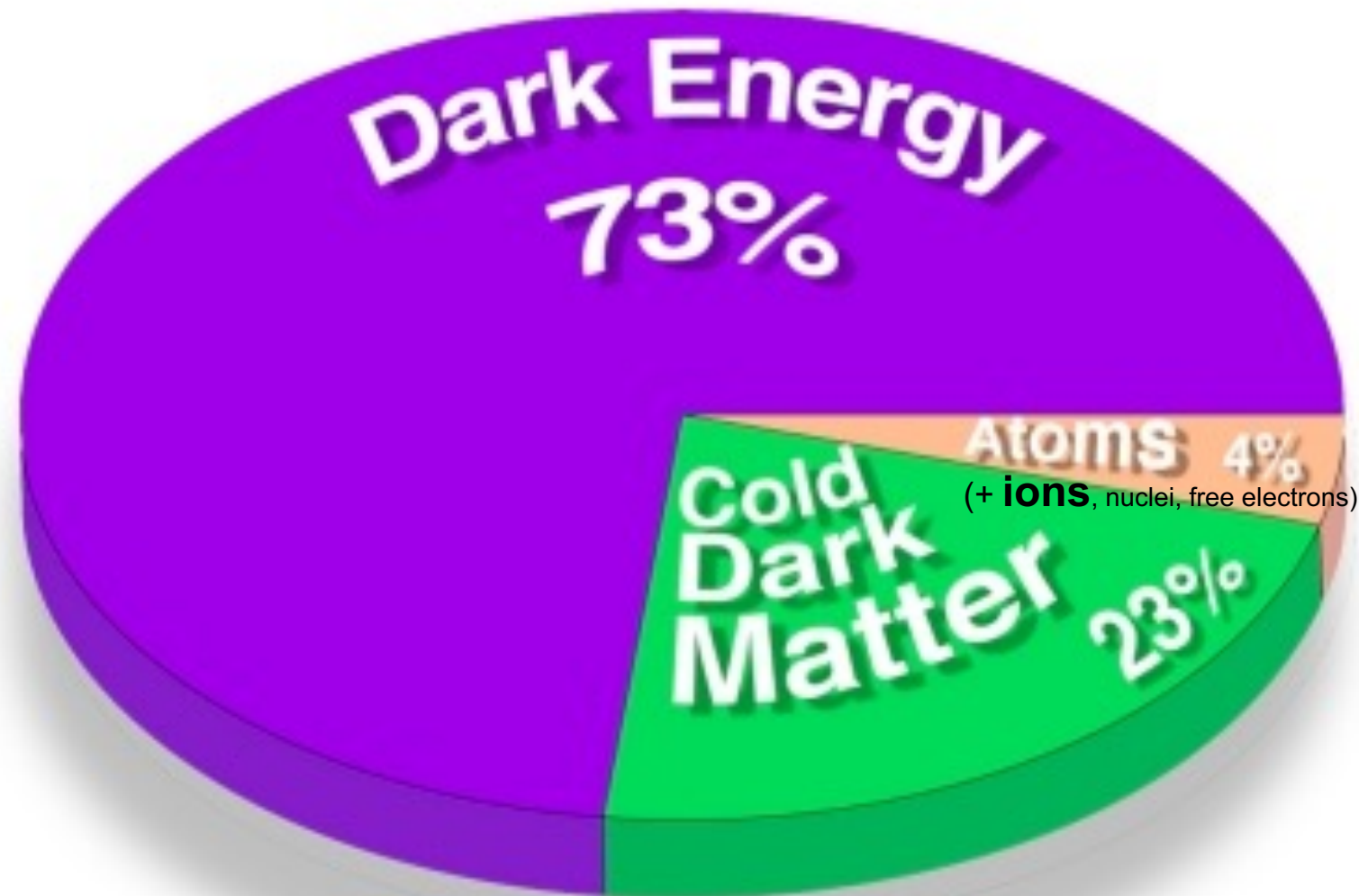
Justin Albert
University of Victoria



jalbert@uvic.ca

Engineering Physics
Seminar

4 Dec. 2025



Also!:

Neutrinos: $\sim 0.15\%$

Electromagnetic radiation: $\sim 0.005\%$

Gravitational radiation: *negligible*

Unstable SM matter & radiation: *negligible*

Antimatter: *negligible*

WHY??

How do we know this??

And what are DE & CDM???

***Space
Expands***

**Edwin Hubble
1929**





Henrietta Swan Leavitt (1868-1921)

Discovered the "Leavitt Law" in 1912 which relates the luminosity vs. the period of Cepheid variable stars ... which then allowed Hubble to determine that the Universe is expanding in 1929. (Leavitt & Pickering, HCO Circular 173, 1 (1912))

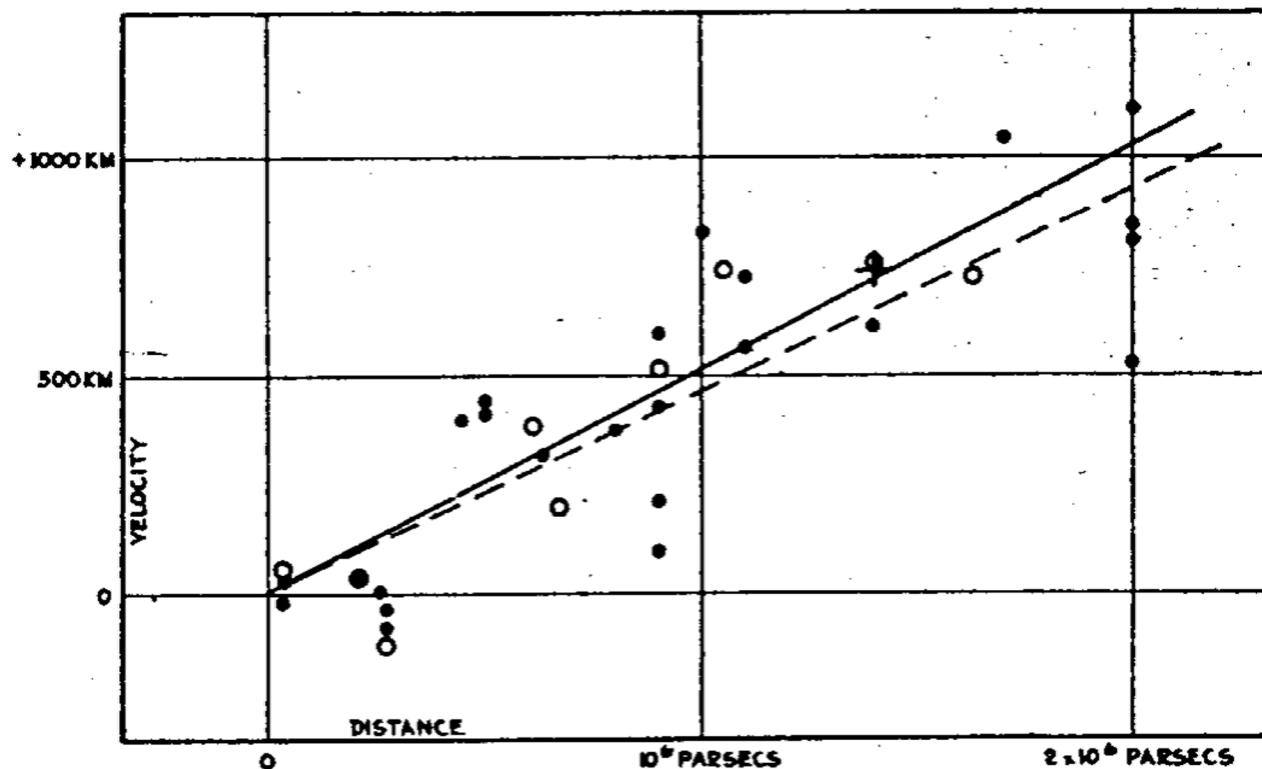


FIGURE 1

Hubble, PNAS 15, 168 (1929)

Einstein's Equations of General Relativity (1916)

Matter tells **space-time**
how to curve:

Space-time tells **matter**
how to move:

**Cosmological
constant**

(energy
density
of empty
space)

Metric
tensor

**Stress-
energy
tensor**

(from
matter &
radiation)

$8\pi G$

c^4

$T_{\mu\nu}$

Einstein, Annalen der Physik **354**, 769 (1916)

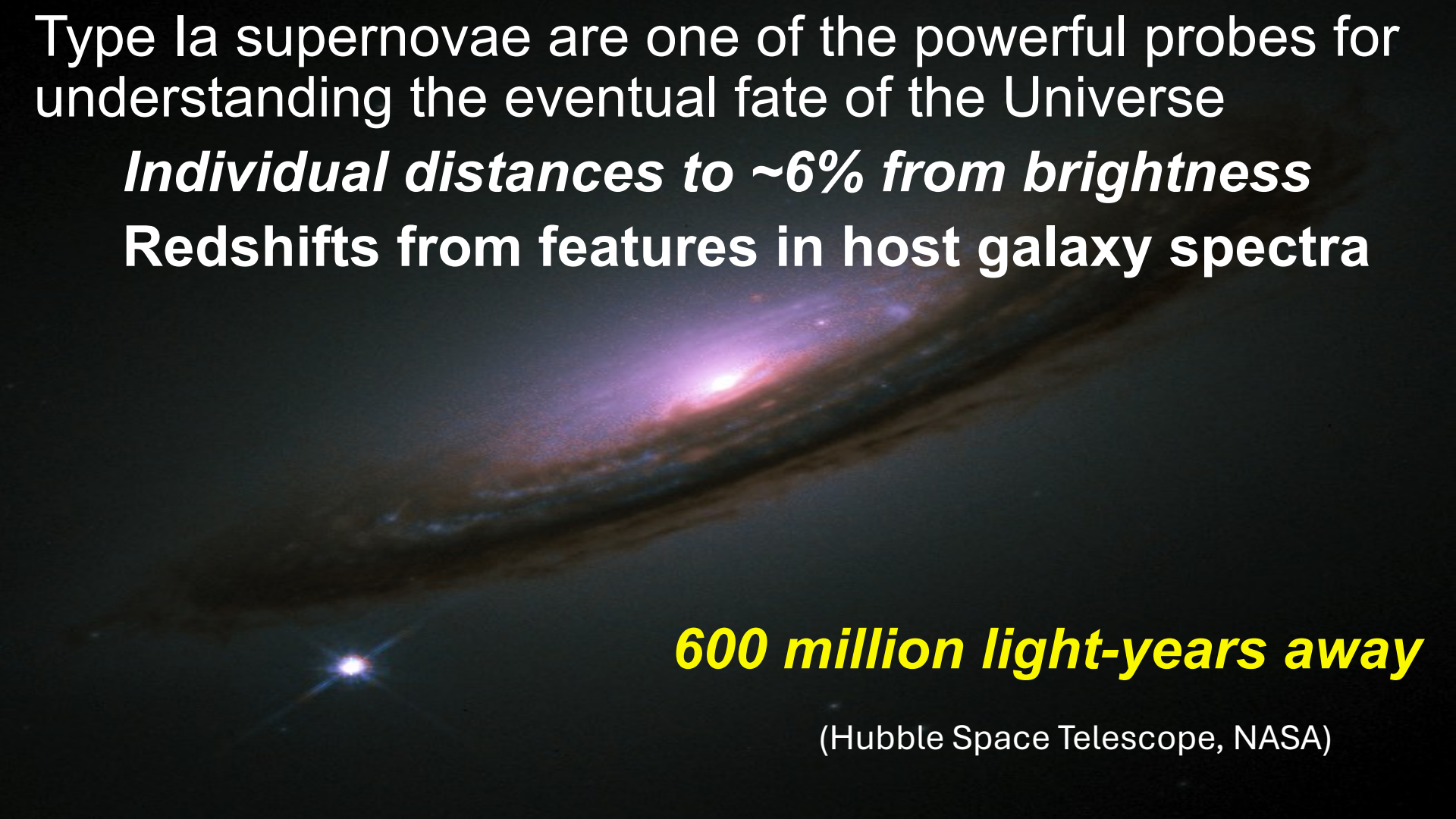
Type Ia supernovae are one of the powerful probes for understanding the eventual fate of the Universe

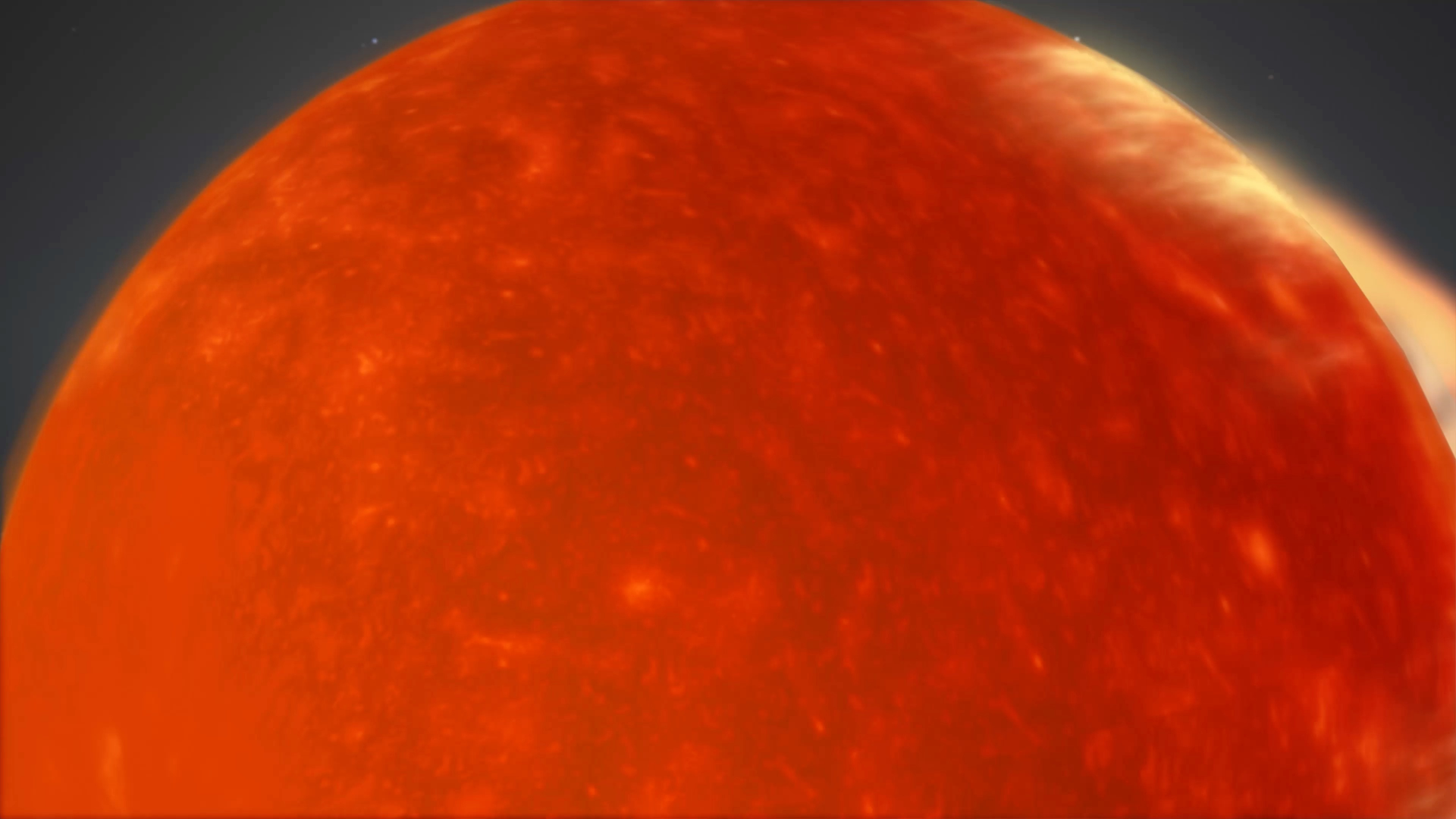
Individual distances to ~6% from brightness

Redshifts from features in host galaxy spectra

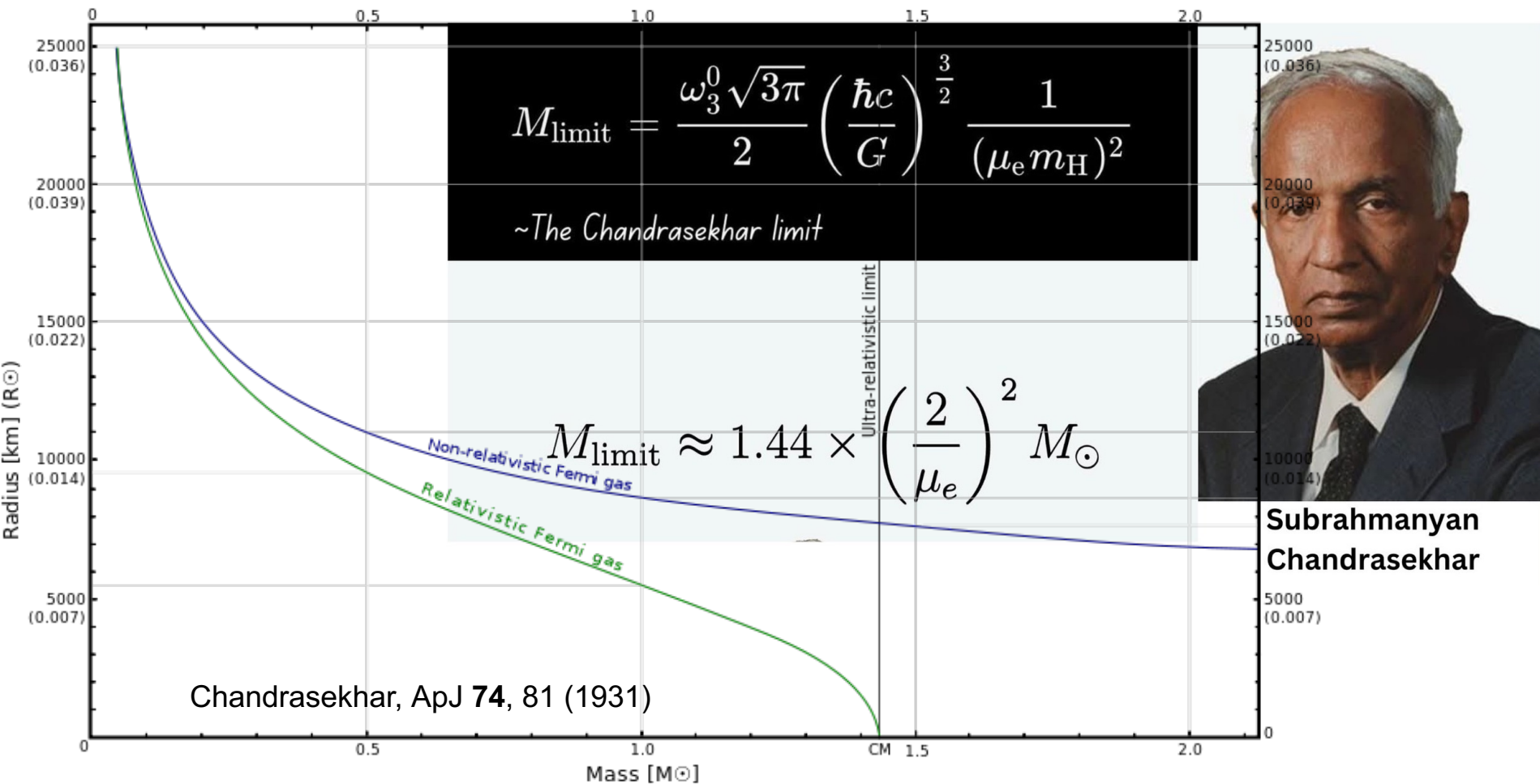
600 million light-years away

(Hubble Space Telescope, NASA)





Chandrasekhar Limit on White Dwarf Mass





Hubble's Law

From redshift and distance measurements, we can express the recession speed v of a galaxy located at a distance d away from us by:

$$v = d \times H_0 .$$

The value of the **Hubble Constant** H_0 is:

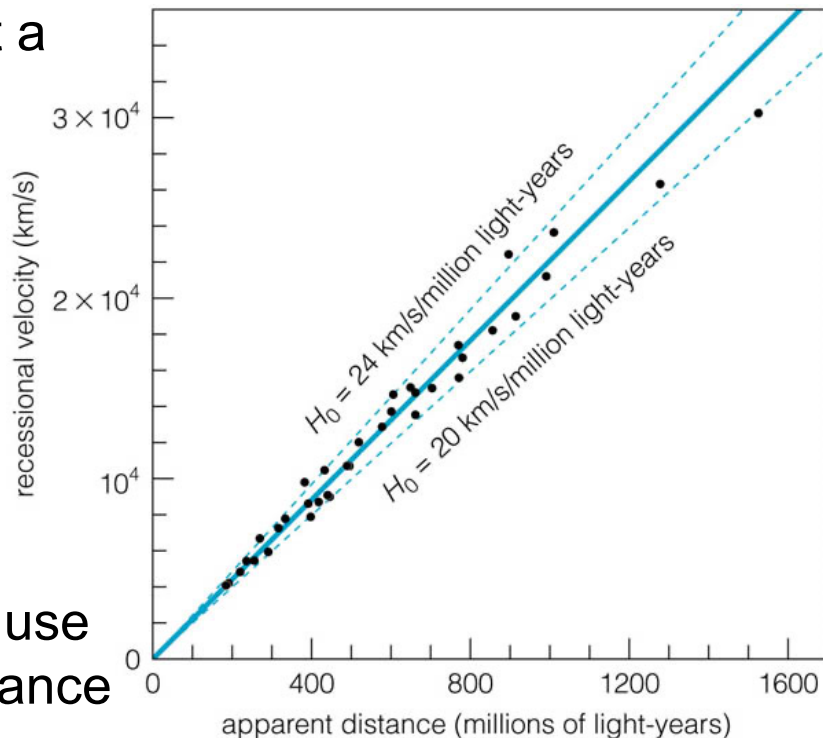
$H_0 = \sim 21.5$ [km/sec] / million light-years

or as more commonly expressed: ~ 70 [km/sec] / Mpc

(1 parsec (pc) = 3.2616 light-years)

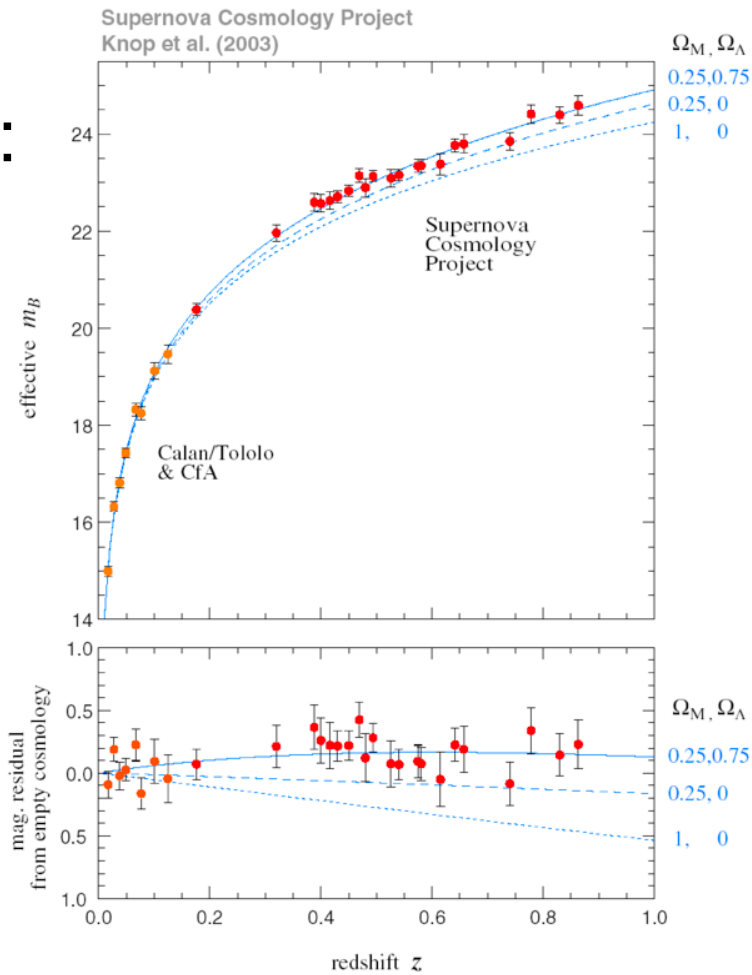
Once the value of H_0 is determined, we can use measured recession speeds to infer the distance of galaxies using the formula:

$$d = v / H_0 .$$



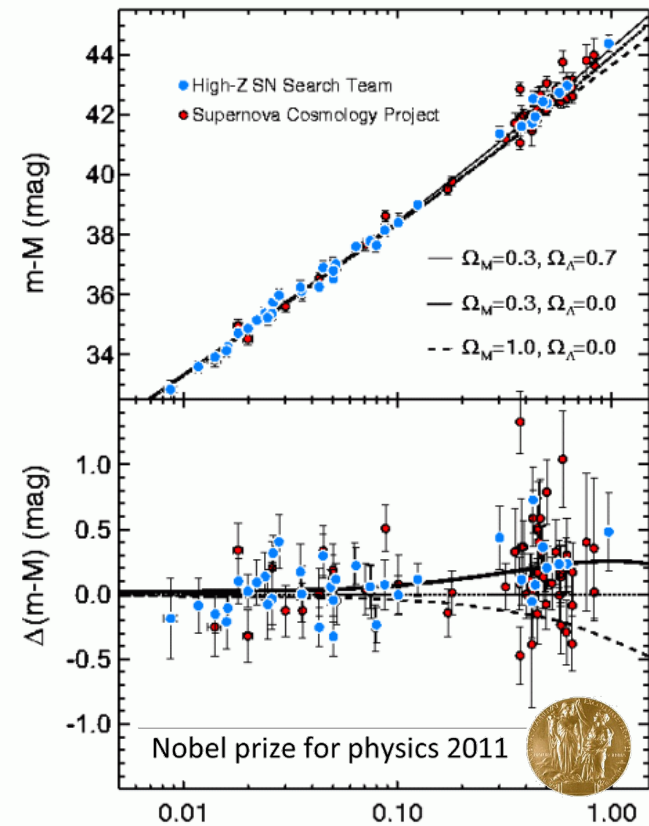
Using SNe Ia to Extend It ...

(as of 2003 ...) :

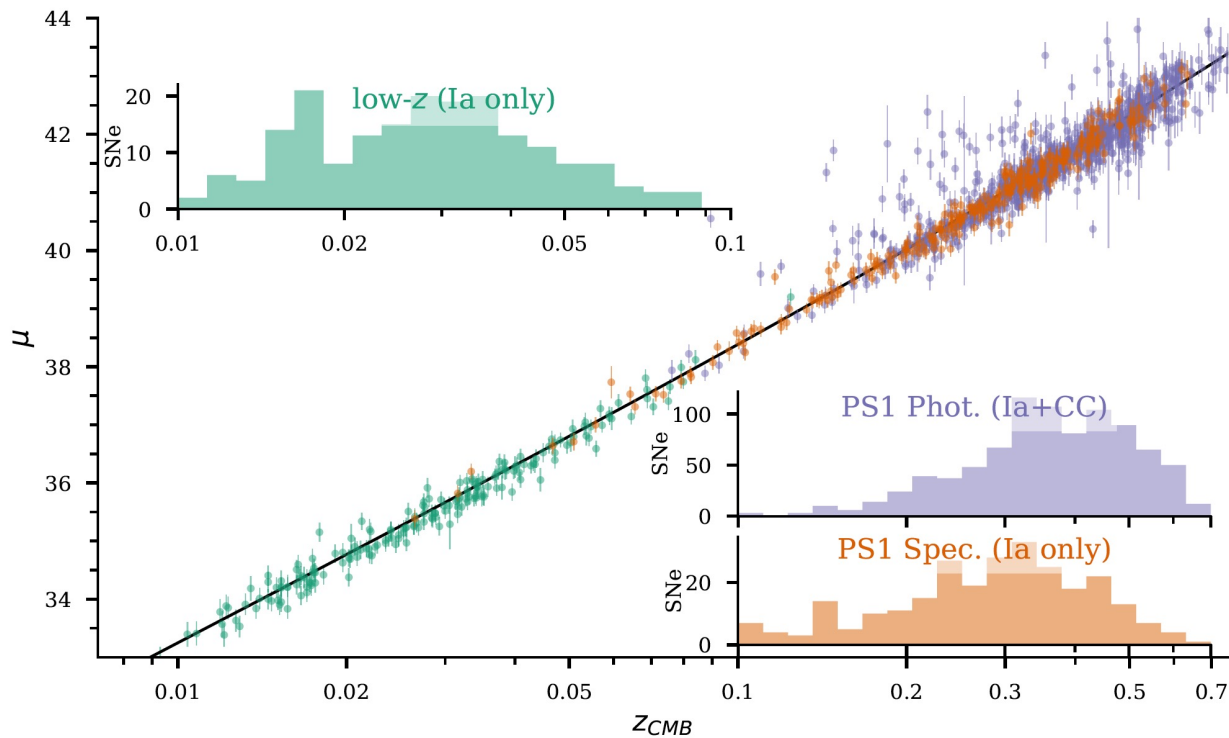


(and 15 years thereafter ... :)

Jones et al. (Pan-STARRS PS1 Survey) 2018, ApJ 857, 51



Nobel prize for physics 2011



Uncertainty on astronomical photometric calibration has **COMPLETELY**
DOMINATED all SNe Ia dark energy measurements for well over a decade now ...

**SNLS: Conley et al
(2011), ApJS 192, 1**

Table 7: Identified systematic uncertainties

Description	Ω_m	w	Rel. Area ^a	w for $\Omega_m=0.27$
Stat only	$0.19^{+0.08}_{-0.10}$	$-0.90^{+0.16}_{-0.20}$	1	-1.031 ± 0.058
All systematics	0.18 ± 0.10	$-0.91^{+0.17}_{-0.24}$	1.85	$-1.08^{+0.10}_{-0.11}$
Calibration	$0.191^{+0.095}_{-0.104}$	$-0.92^{+0.17}_{-0.23}$	1.79	-1.06 ± 0.10
SN model	$0.195^{+0.086}_{-0.101}$	$-0.90^{+0.16}_{-0.20}$	1.02	-1.027 ± 0.059
Peculiar velocities	$0.197^{+0.084}_{-0.100}$	$-0.91^{+0.16}_{-0.20}$	1.03	-1.034 ± 0.059
Malmquist bias	$0.198^{+0.084}_{-0.100}$	$-0.91^{+0.16}_{-0.20}$	1.07	-1.037 ± 0.060
non-Ia contamination	$0.19^{+0.08}_{-0.10}$	$-0.90^{+0.16}_{-0.20}$	1	-1.031 ± 0.058
MW extinction correction	$0.196^{+0.084}_{-0.100}$	$-0.90^{+0.16}_{-0.20}$	1.05	-1.032 ± 0.060
SN evolution	$0.185^{+0.088}_{-0.099}$	$-0.88^{+0.15}_{-0.20}$	1.02	-1.028 ± 0.059
Host relation	$0.198^{+0.085}_{-0.102}$	$-0.91^{+0.16}_{-0.21}$	1.08	-1.034 ± 0.061

^aArea relative to statistical only fit of the contour enclosing 68.3% of the total probability.

Note. — Results including statistical and identified systematic uncertainties broken down into cat
In each case the constraints are given including the statistical uncertainties and only the stated sys
contribution. The importance of each class of systematic uncertainties can be judged by the relat
compared with the statistical-only fit.

Basic Technique & Goal: *A 0.1% Calibrated, Mobile Source Above the Atmosphere*

Balloon Payload

100 mW lasers

440 nm

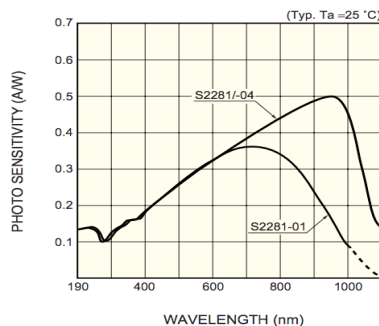
532 nm

658 nm

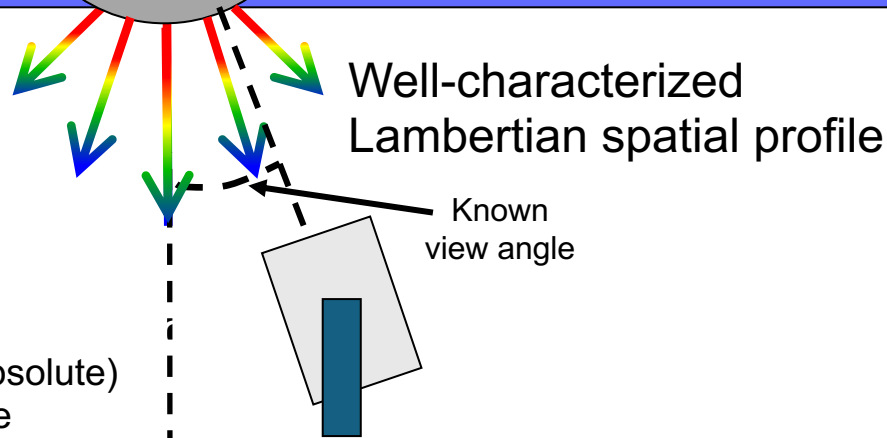
830 nm

Integrating sphere

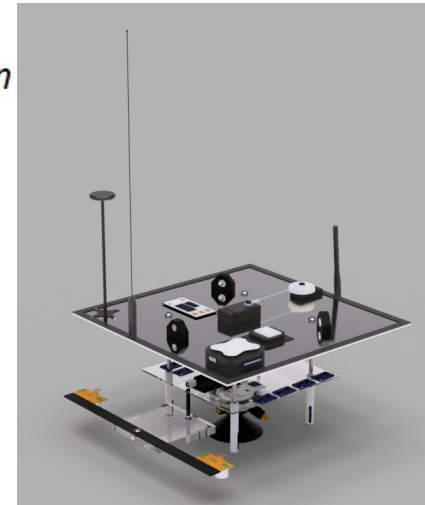
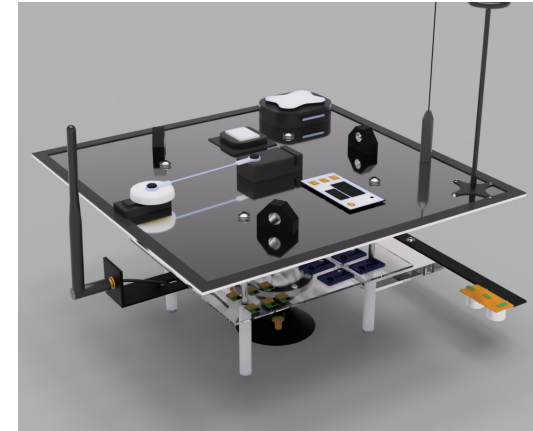
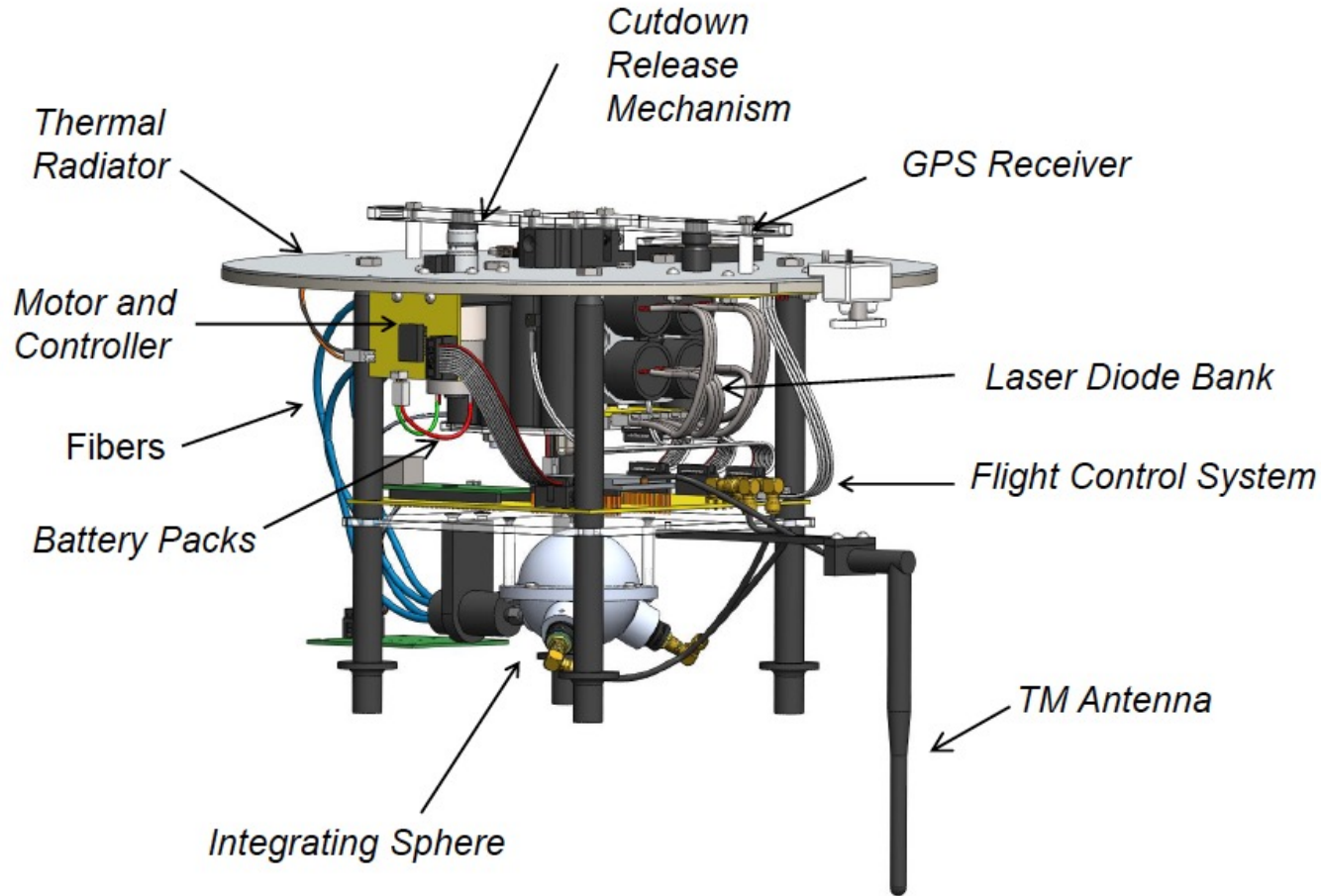
NRC- & NIST-calibrated photodiodes



NRC- & NIST-calibrated ($\sim 0.1\%$ absolute)
photodiode spectral response

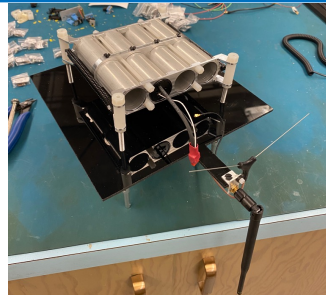
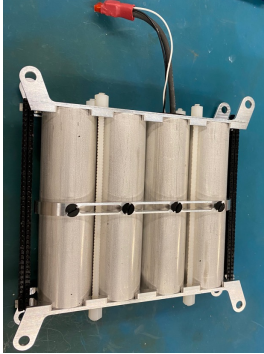


Basic Payload Design for Photometric Calibration



Light Sources & Calibrated Monitoring Photodiodes on 2" ID Integrating Sphere

Four WorldStarTech 100 mW laser diode modules, respectively **440 nm**, **532 nm**, **658 nm**, & **830 nm**, each fiber coupled into a fused-ends, "quadfurcated" MM optical fiber (custom fab'd by Ceramoptec).



Can be swapped out for **635 nm** or **780 nm**.

Can be swapped out for **808 nm** or **850 nm**.

Electronics
MCU, TIA,
Laser drivers

Photodiode 1

Photodiode 2

Integrating
Sphere
(coated with
Avian B)

Can be swapped out for **405 nm**.

Can be swapped out for **520 nm**.

Laser Module

One photodiode is a **Hamamatsu S12698-01 (Si)** and the other is a **Hamamatsu G10899-03K (InGaAs)**.

(Utilizing two different photosensor material technologies is a useful cross-check.)



Photodiode readout and digitization

BIAS SWITCH
Voltage divider for the photodiode bias
 $V = R_{10} / (R_{10} + R_{11}) \cdot V_{DD}$
(Resistance at SET pin)

Logic LO = Disengaged

For flight, populate SET resistor with appropriate value, and remove trimmer pot. SET resistor should be low drift.

Larger capacitance on SET pin reduces output ripple level.

Current Limit Calculation:
 $I_{lim} = 150 / R_{10} \text{ [A]}$
(Resistance at ILDM pin)

Max output current is 500mA. To remove current max setpoint, the ILDM to GND.

Max output current is 500mA. To remove current max setpoint, the ILDM to GND.

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Max output current is 500mA. To remove current max setpoint, the ILDM to GND.

GAIN SW1

GAIN SW2

TIA POSITIVE REFERENCE

Even though the output caps are tied directly to the photodiode cathode, they do not unbalance the system. This has been checked in TINA-TI.

PGFIB threshold = $0.1 \cdot (1 + R_{14}R_{15}) + (25\text{mA} \cdot R_{14})$

Set the cutoff voltage for fast start

PD_K

BIAS SW

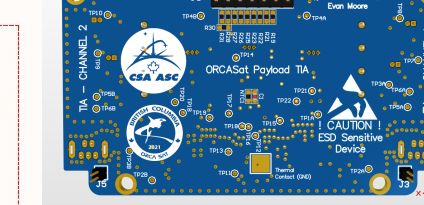
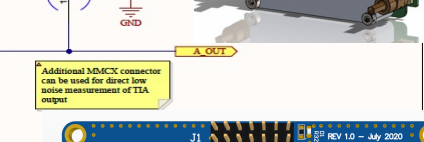
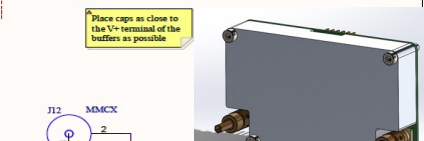
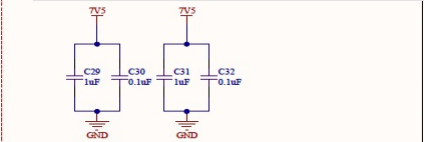
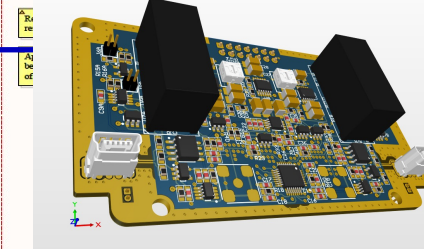
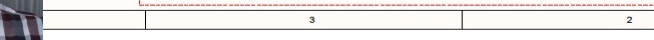
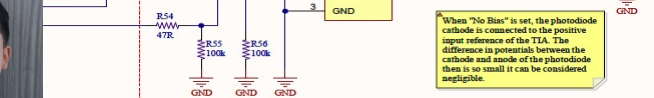
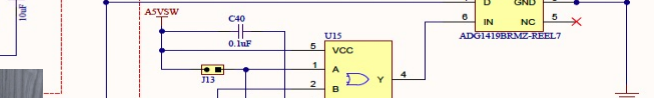
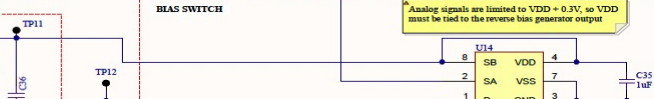
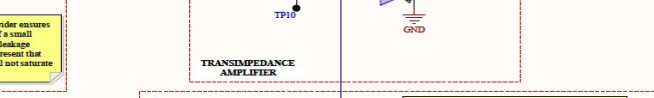
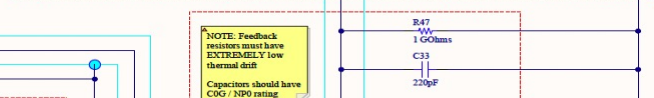
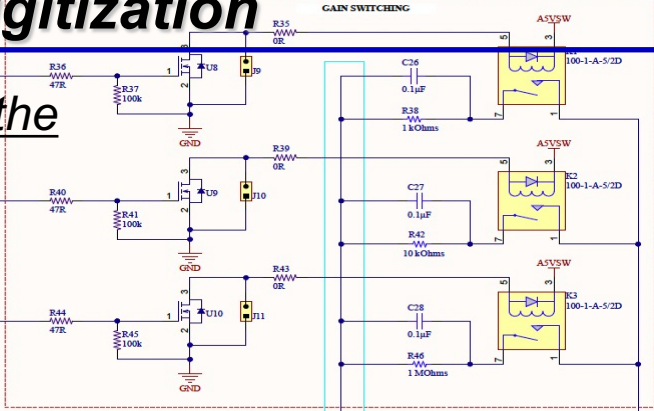
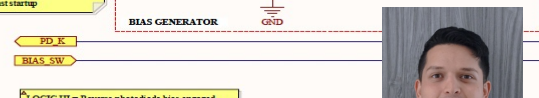
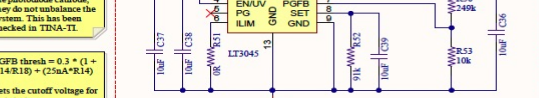
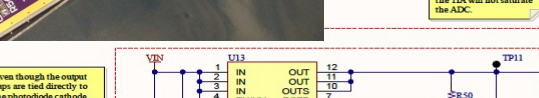
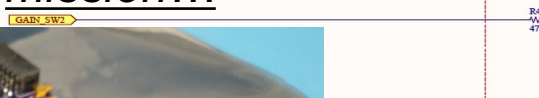
LOGIC HI = Reverse photodiode bias engaged

LOGIC LO = No bias on photodiode

LOGIC LO = No bias on photodiode

LOGIC LO = No bias on photodiode

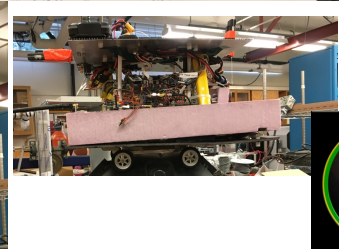
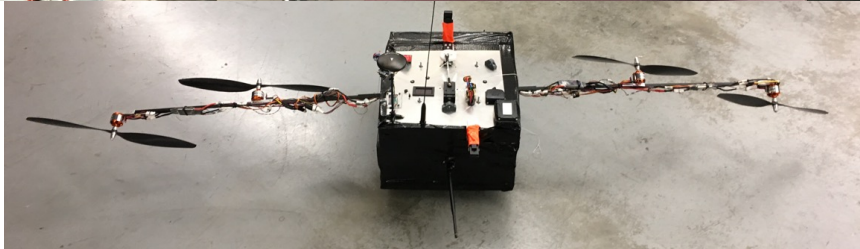
LOGIC LO = No bias on photodiode



Title	Number
Size	Tabloid
Date	10-13-2021
File	c:\Users\TIA_SchDoc
Sheet of	Drawn By



Payload

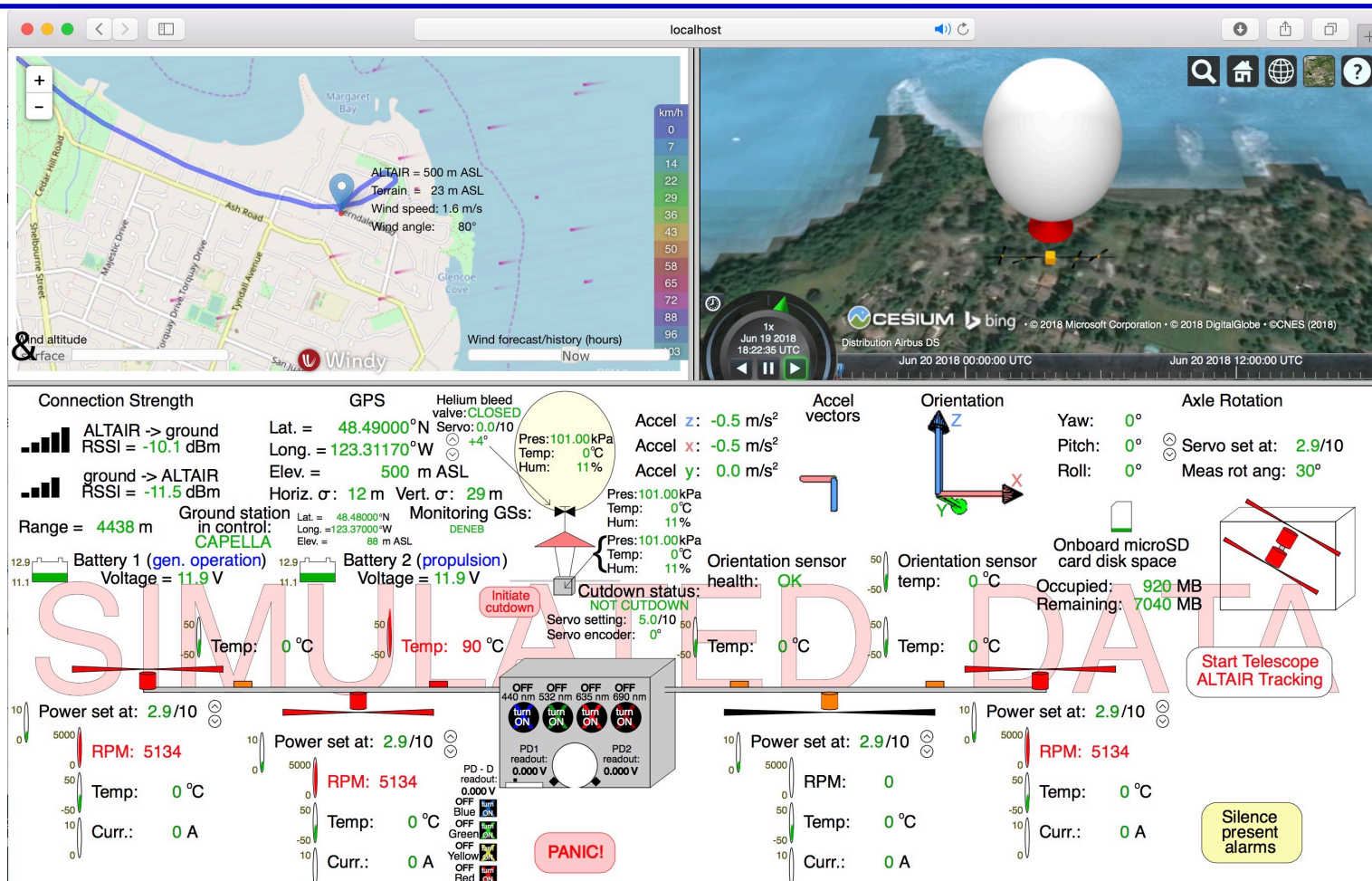


Flight control – AIFCOMSS

Code actively maintained / updated at:

<https://github.com/ProjectALTAIR/AIFCOMSSwithCUPredictorTest>

ALTAIR
Integrated
Flight
Control,
Operation,
Monitoring, &
Simulation
System





ALTAIR 12

2014.05.18 05:00:24

Onboard transponder (uAvionix ping900XR)

For continuous real-time communication with and identification to air traffic control.
(FAA & Transport Canada certified.)



So that we don't get shot down by NORAD!!! ;)
(Or by the Chilean air force!!!)



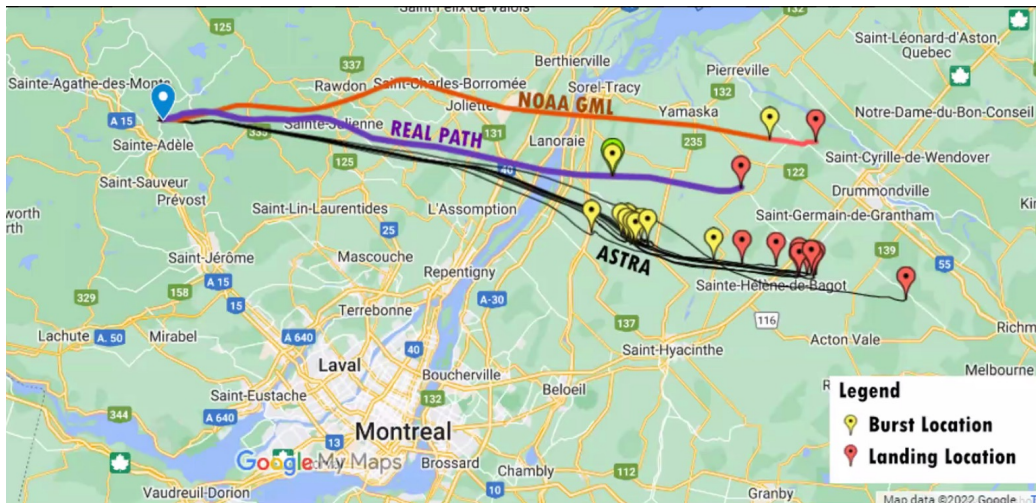
Vera Rubin Observatory
on Cerro Pachon in Chile



Operations & McGill / Montreal Test Flights



910 MHz
portable
directional
ground
antennas,
range approx.
100 km.



Always ≥ 2
ground
stations in
contact.

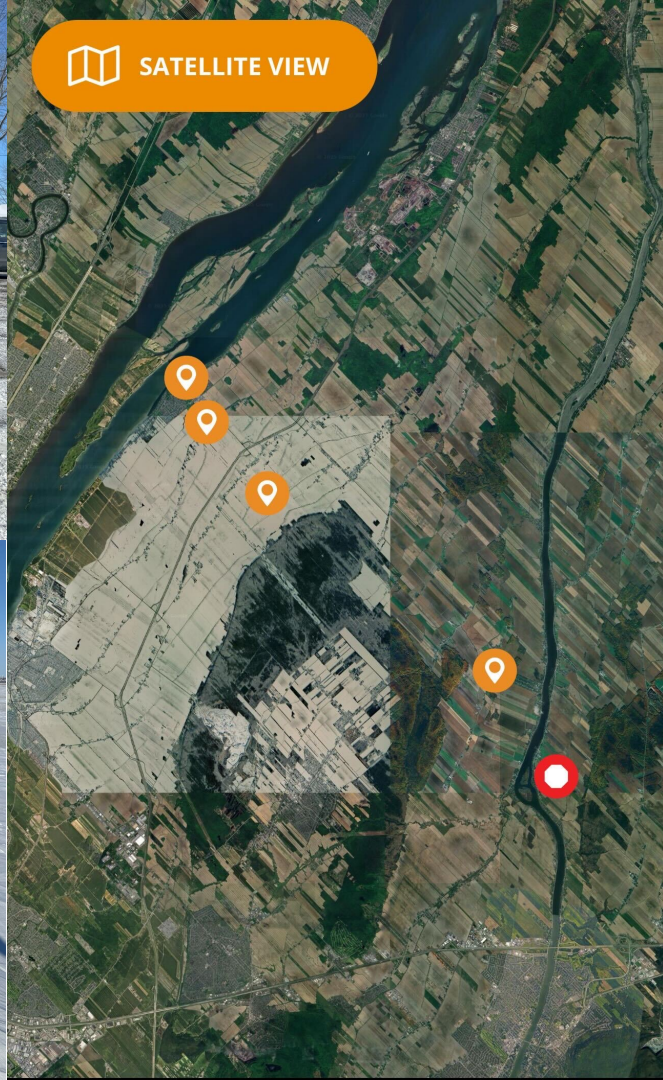


Daytime Flight
Sat. Feb. 1, 2025:

Inflation
(launch from
Verchères, QC
11:45 am
local time):



Payload
recovery:



Flight path
←

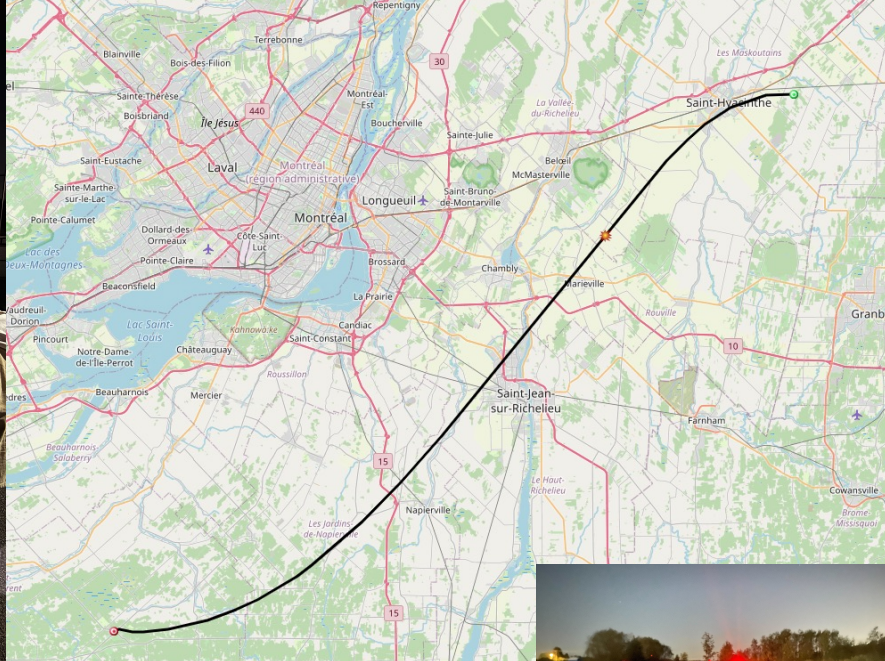
Flight
duration
45 mins



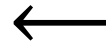
Nighttime Flight

Sun. Sept. 7,
2025:

Inflation
(launch from
Saint-Antoine-
Abbe, QC
9:30 pm
local time):



Flight path



Flight
duration
~1 hr

Our ground
telescope
observations
of this
particular
flight were
unfortunately
obscured by
unpredicted
low clouds:



Payload
recovery
~10 am the
following
morning:



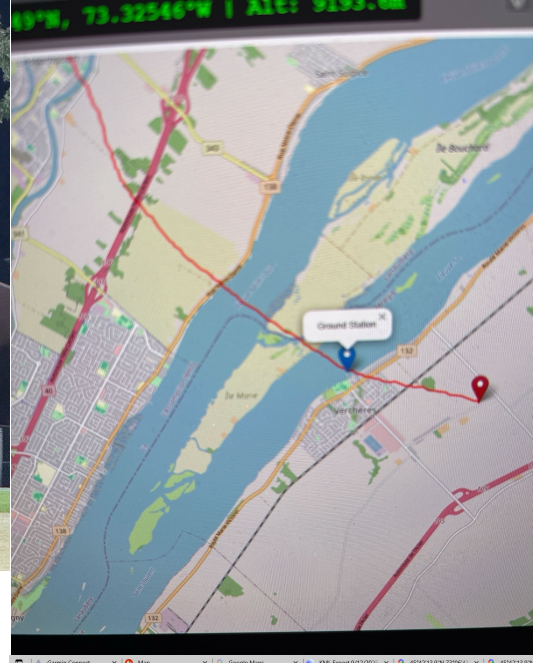
Nighttime Ground- Observations Flight

Thu. Sept. 11,
2025:

Inflation
(launch from
Parc des Moissons,
L'Assomption QC
9:50pm local time):



Observations
from Pioneer
Park, Verchères:

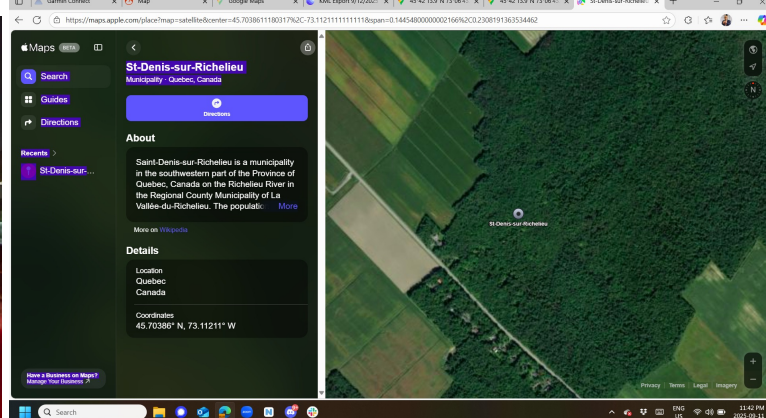


Flight path



Flight
duration
45 mins

Payload
landing in
trees near
St-Denis-
sur-





Payload recovery
mid-day on the
following day,
Fri. Sept. 12,
2025:

(just after noon at
~12:10 pm local
time):



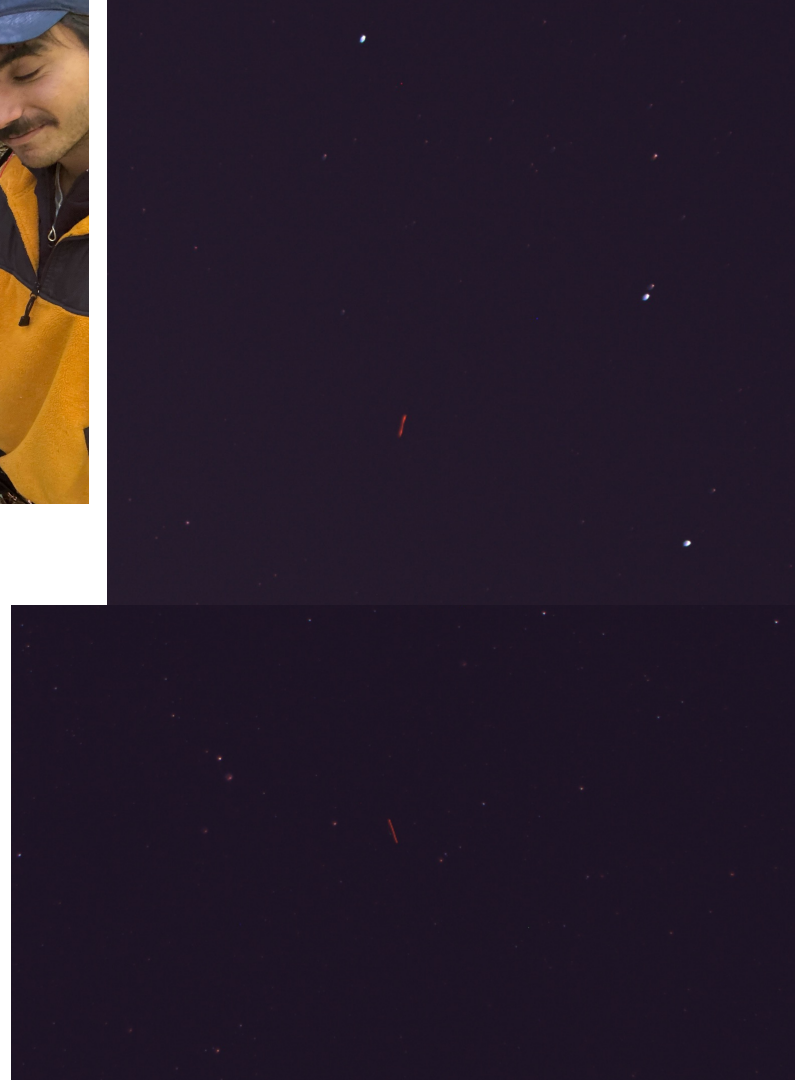
... utilizing our
arborist's sling:



A couple of
our payload
ground-
telescope
observation
images from
this flight



Image
analysis
ongoing;
we're
scheduled to
report our
observation
results, with
measured
photometric
precision
from this and
our other
flights, at AAS
in Phoenix on
6 Jan. 2026.



Conclusions

- Artificial sources can in principle provide up to two orders of magnitude better photometric calibration precision than any stellar, non-solar light sources can presently.
 - 1) Can *study them right in the lab before and after use*, unlike stars.
 - 2) Can *monitor them in-situ*, in real time.
 - 3) Can also be used to *calibrate white dwarfs* (and other stars, and the Moon) very precisely, and on a detector-based standards scale.
 - 4) Small balloons are *inexpensive!!!* E.g.: Rubin could own 3 or so!!! – launching them at dusk each night, flying to different areas of the sky, and having them *return home* at dawn *each morning!*
 - 5) *Fully observer-controlled choice of spectrum* & color on demand, ... and *brightness*, ...exact *location* in the sky as function of time of night (or day), ...
- **Not restricted to optical/IR!**: e.g. a precisely-polarized Gunn diode-based precision microwave polarimetry calibration source could also be flown. (Radio, near-UV...)
- Balloons, of course, do not replace additionally having a satellite source to (at least occasionally!) determine effects from the very top of the atmosphere!!! But ALTAIR-style balloons can (and I posit will!) be the low-cost, ground-based-observatory-owned, workhorses.



Some (very important!!!) separate & additional conclusions

- A very interesting (& very useful) fact about Earth's atmosphere is that, at mid-latitudes, winds in the upper troposphere statistically tend to be in opposite direction to winds in the lower stratosphere.
- Thus, with just very small amounts of (weather-optimized) upward and downward propulsion (just above and below the tropopause), powered using very lightweight flexible solar panels on top of the balloon (+ onboard rechargeable LiPoly batteries for nighttime power), one could utilize the winds to **circulate near the tropopause over a given region of Earth** (e.g., over a region the size of Illinois, or of Vancouver Island, etc).
- After **a few weeks**, a significant fraction of the helium will diffuse through the balloon latex, and the balloon would **descend to base** for (simple and low-cost!: latex balloons are about \$150) balloon and helium replacement. Then, balloon and payload would be sent back up again.
- This opens up many additional, very different usage possibilities:
 - 1) **Communication**: replacing the need for cell-phone towers with very wide coverage balloons in remote (and/or not-so-remote) areas.
 - 2) **Earth observation** (for, e.g., long-term studies of Arctic ice, etc.)
 - 3) !

ORCASAT

- Lessons Learned from the First Optical Source Satellite
- Mission for Calibration of Ground-Based Observatories

An initial LEO testbed for goal of precision (~0.1% level) calibration of astronomical photometry / magnitudes / flux at **660 nm** and **840 nm** wavelengths.

Nov. 26, 2022: Launch from KSC to ISS

Dec. 29, 2022: Deployment out of ISS → orbit

July 7, 2023: Atmospheric re-entry, burn up

GRS-26 launch

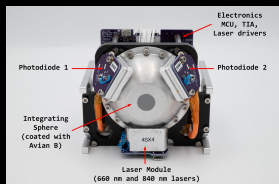
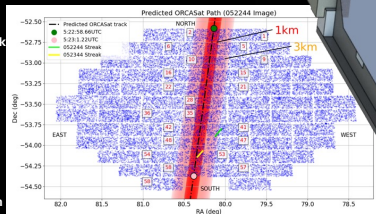
Nov. 26, 2022
(ORCASat one of many items aboard Cargo Dragon atop Falcon-9)

We captured this image using DECam on 31 Jan. 2023, with a streak light in RA-dec where ORCASat should have been & exactly when the light source was. Was it ORCASat...?

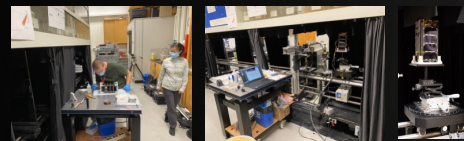
This streak (image taken with DECam on 4 Mar. 2023) really definitely was ORCASat!!! But unfortunately what it wasn't is the ORCASat onboard light source... The image was taken only 1 hour after sunset on Cerro Tololo, when ORCASat was not yet in Earth's shadow, and the streak is reflected sunlight from the spacecraft surface.

Deployment
Dec. 29, 2022

from Nanoracks
deploy module
405 km alt.



Pre-launch payload light source calibration at NRC Ottawa during summer 2022:



Main Lessons Learned:

Future small, low-cost photometric calibration satellites in LEO can/would/will be useful!!! Calibration with streaks (rather than or in addition to point sources) is OK!!! However!!!:

- 1) Need at least approx. **10x brighter** (& 1 or 2 more, if possible, for > wavelength range) onboard light sources!!! I.e., $\geq \sim 1$ watt, i.e. 10x brighter than ORCASat's ~ 100 mW, sources would be sufficient!!!
- 2) Needs **well-functioning ADCS!** The attitude determination and control system (ADCS) that we had purchased from CubeSpace had fundamental flaws, e.g. an unshielded magnetometer cable.
- 3) Needs just a **slightly higher LEO orbit**, which won't decay in 6 months!!!! (Just **10 km higher** would have meant a multi-year mission!!!!...)

Poster prepared by
Diego Reyes and
Justin Albert

University of Victoria
October 24, 2024



Mission was funded by:
with gracious
support from additional sources.



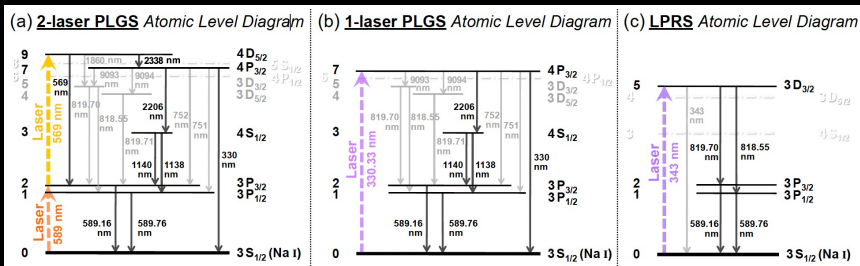
<https://www.orcasat.ca>

Laser Photometric Ratio Star (LPRS):

An **Ultra-Precise** ($< 10^{-4}$) Spectrophotometric Calibration Ratio, using a Novel Excitation of the Earth's Sodium Layer

Could one create a bi-chromatic laser guide star, utilizing a **cascade decay** of a laser-excited component of the upper atmosphere, in which one would be assured that, to **extremely high precision**, the numbers of photons generated in the two colors are equal, due to the cascade nature of such a decay...?

Yes! – but not in the way that polychromatic laser guide stars have ever been created or tested so far! Must use the $3D_{3/2}$ or $3D_{5/2}$ excitation of Na I! – no other excitation would work for this in Earth's atmosphere!

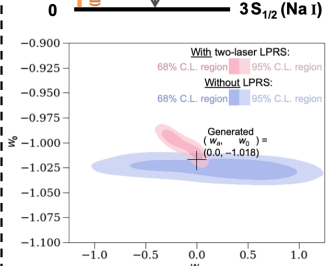
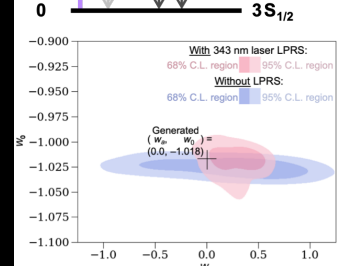
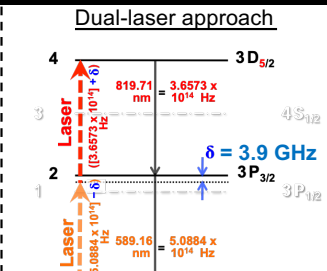
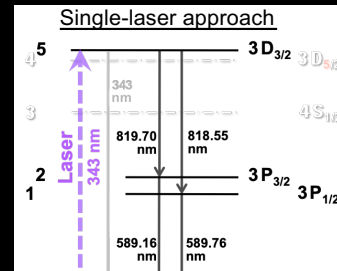
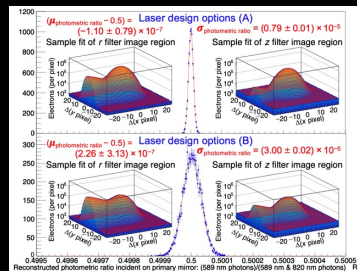
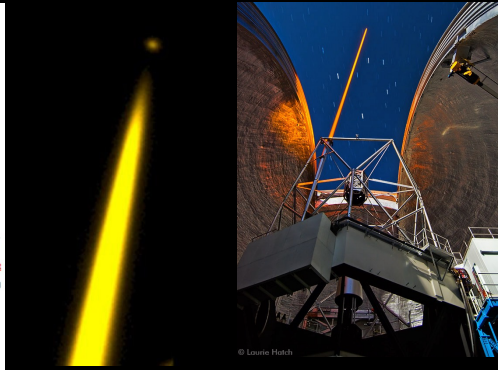
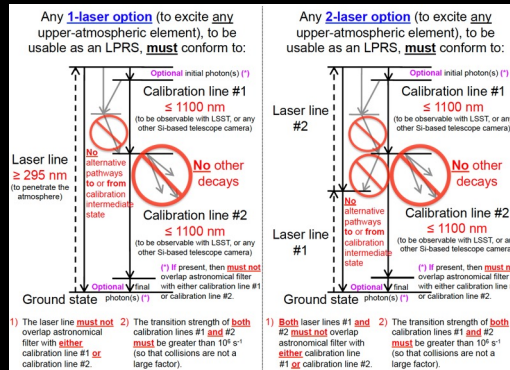


Please kindly see !!! :

Albert et al., MNRAS **503**, 4399 (2021) ;

Albert et al., MNRAS **503**, 4412 (2021) ; Poster prepared by

Albert, Budker, & Sadeghpour,
Nat. Sci. **2**, e20220003 (2022).



I acknowledge and respect the Lək'wəŋən (Songhees and X̱wsep̓səm/Esquimalt) Peoples on whose territory the University of Victoria stands, and the Lək'wəŋən and W̱SÁNEĆ Peoples whose historical relationships with this land continue to this day.

L'Polytechnique Montréal reconnaît qu'elle est située en territoire autochtone non cédé par voie de traité, et souhaite saluer ceux et celles qui, depuis des temps immémoriaux, en ont été les gardiens traditionnels. Elle exprime son respect pour la contribution des peuples autochtones à la culture des sociétés ici et partout autour du monde.



A DUTY TO REMEMBER

36th COMMEMORATION OF THE ÉCOLE POLYTECHNIQUE MASSACRE